



GSFC Flight Network Summary Space Internet Workshop #5

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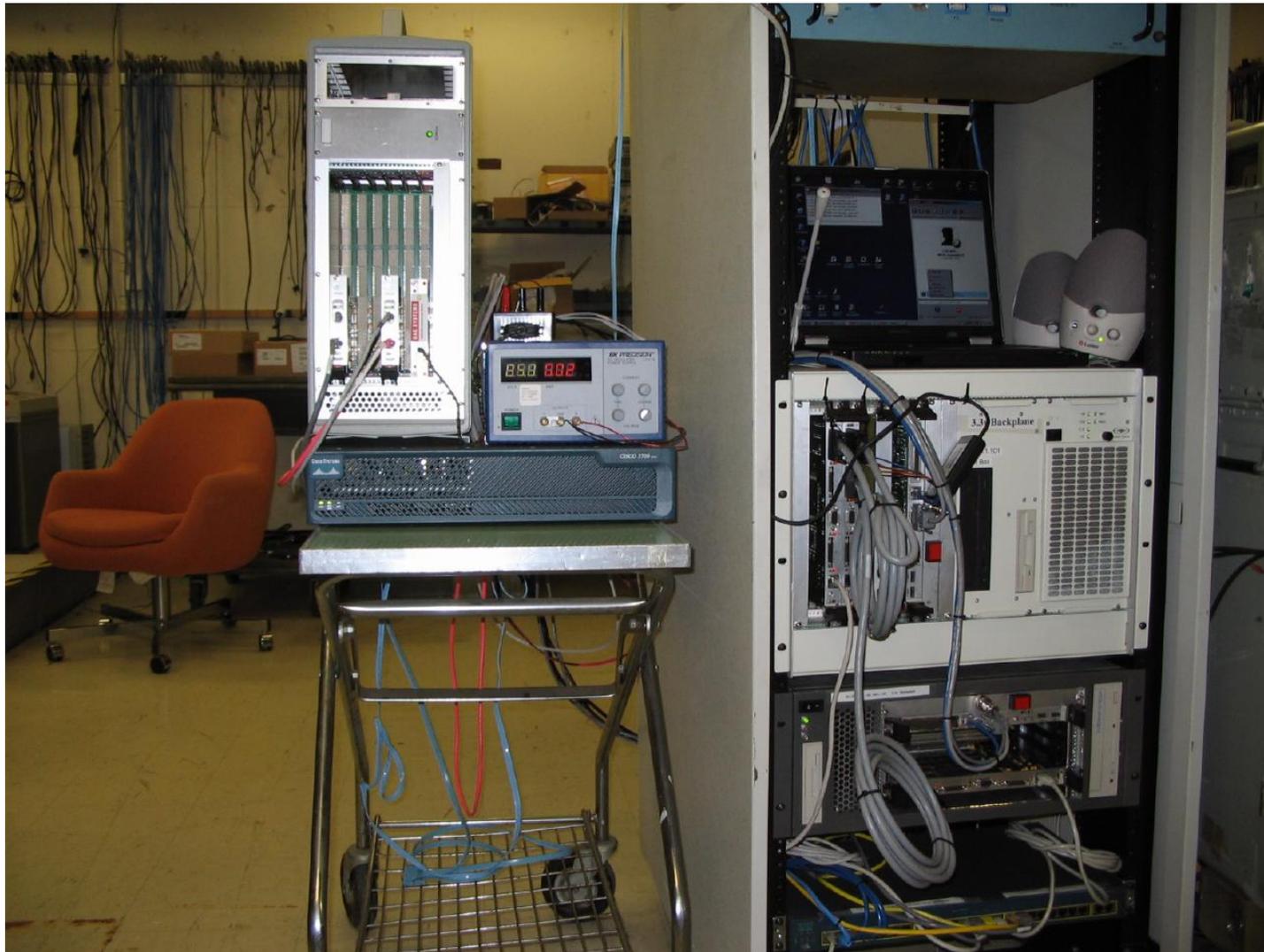
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Cisco Embedded Space Router

- Space Act Agreement between Cisco and GSFC, 2004
- GSFC and Cisco base-lined requirements
- Cisco designed and implemented an IOS on a BAE Systems Rad750; called “SQR” (Space Qualified Router)
 - Clock not specified, somewhere between 133 and 166mhz
 - 128Mb RAM
 - Running experimental 12.3 IOS w/ K9 advanced security option
 - Two commercial dual 10/100/1g Ethernet devices on the CPCI backplane, giving the SQR four Ethernet ports, appearing in IOS as conventional Gigabit Ethernet devices.
 - No onboard flash boot, firmware loaded from auxiliary system.
- GSFC tested router in flight software lab in April 2005
- Demos in June 2005

Cisco Embedded System Router

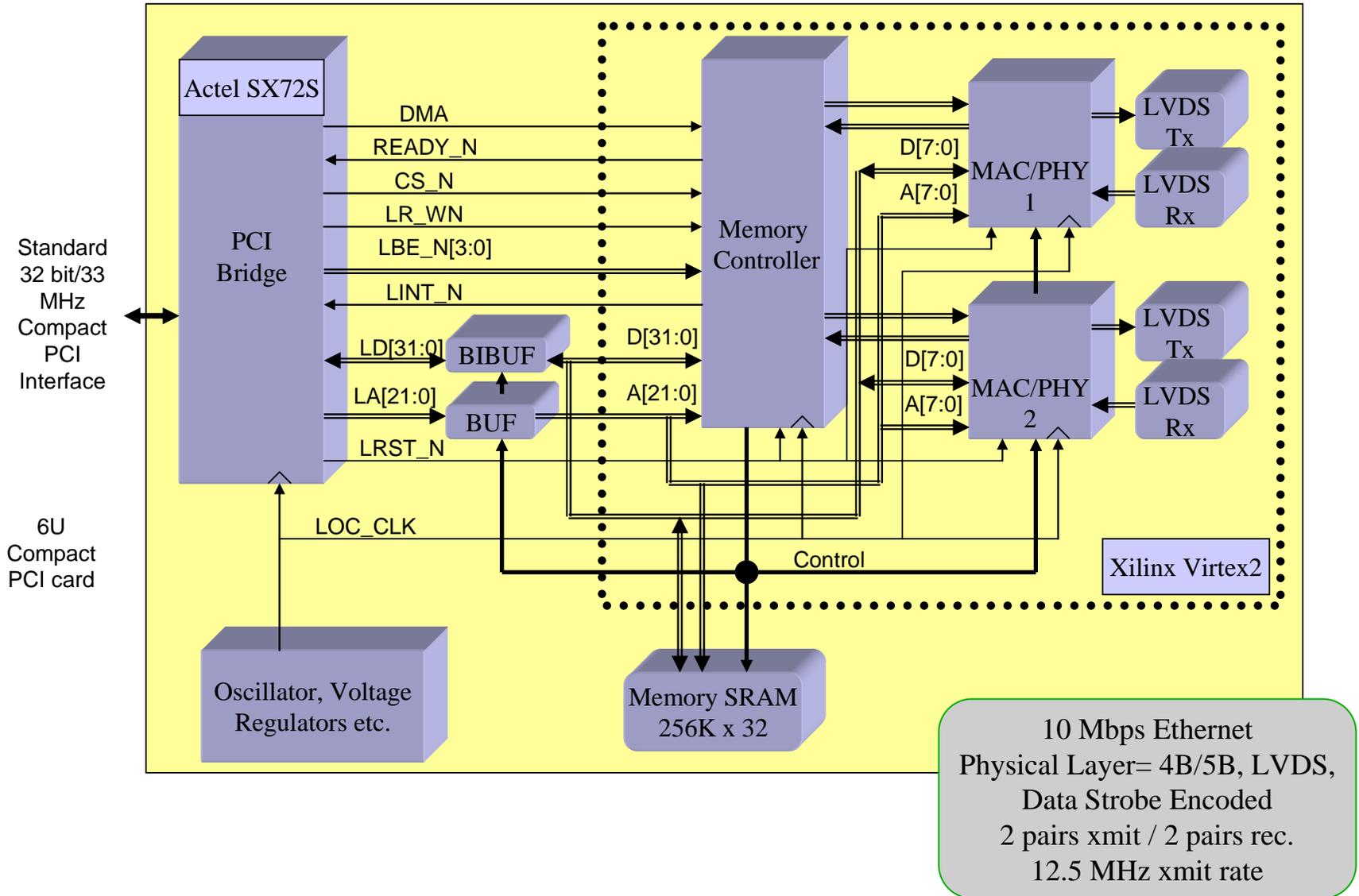




GSFC NIC and Switch

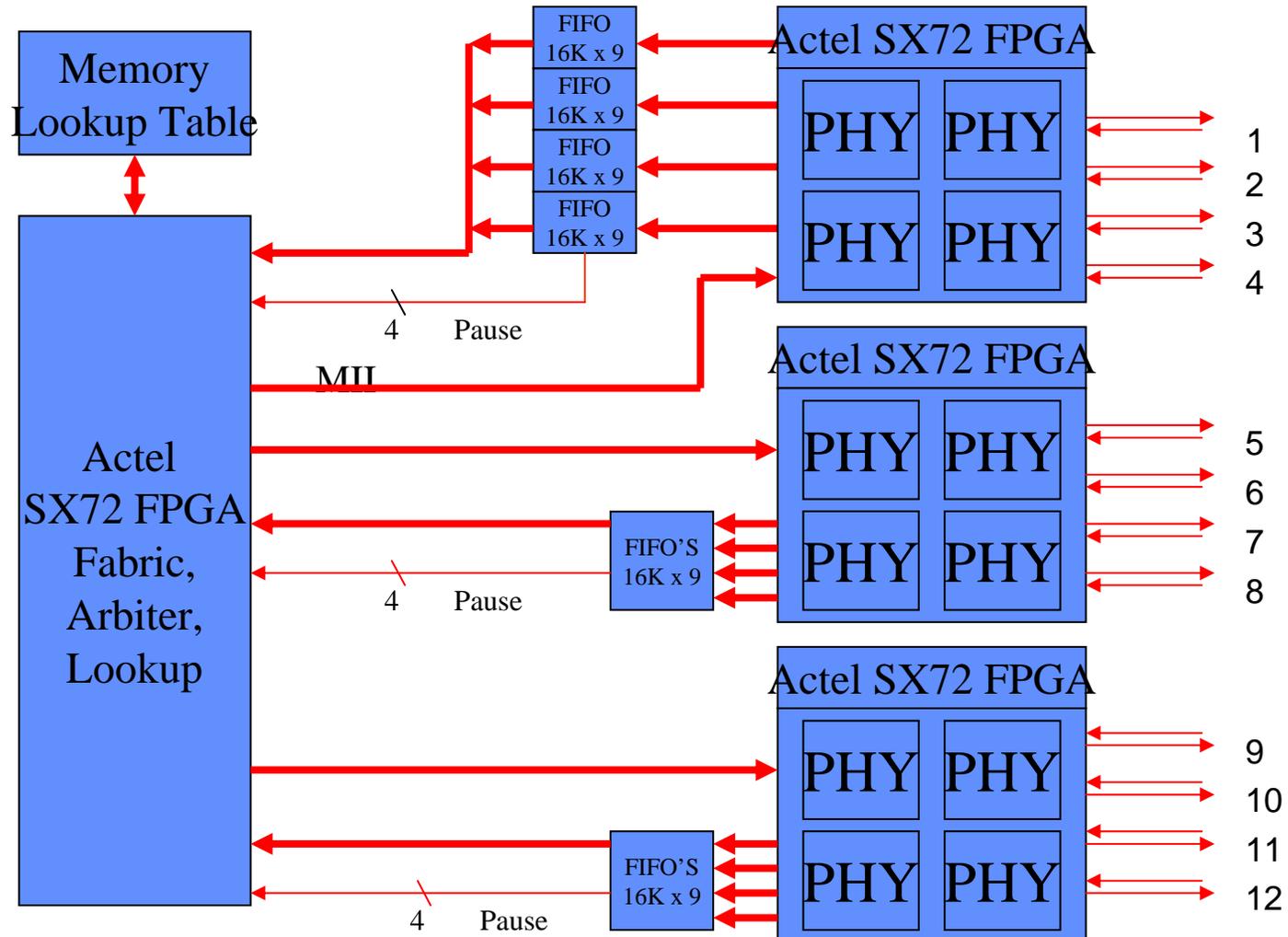
- Developed by Code 561 (Mike Lin) under ESTO funding
- Supports 10/100 Mbit Ethernet over twisted pair using a 12.5/125 Mbit DS Link encoded LVDS physical layer
- Media converter connects LVDS Flight Ethernet to 10/100-base TX
- NIC
 - 2 independent instances of a commercial MAC core per NIC.
 - Two external LVDS Ethernet interfaces & one selectable 10baseT RJ45 interface per NIC. MAC's are standard, PHY redeveloped to use LVDS.
 - NIC configured as a 6U CPCI card; 32 bit, 3.3v, 33mhz, bus mastering.
 - Supports full duplex 10/100 Mbit Ethernet
 - FPGA's chosen to provide a path to a rad-hard flight implementation
- Switch
 - 12 port, 10mb LVDS Ethernet
 - Fixed MAC address table (to reduce gate count)
 - Supports Broadcast and Pause (to meet GPM network requirements)

NIC Breadboard





Rad-Hard Switch





Router Configurations

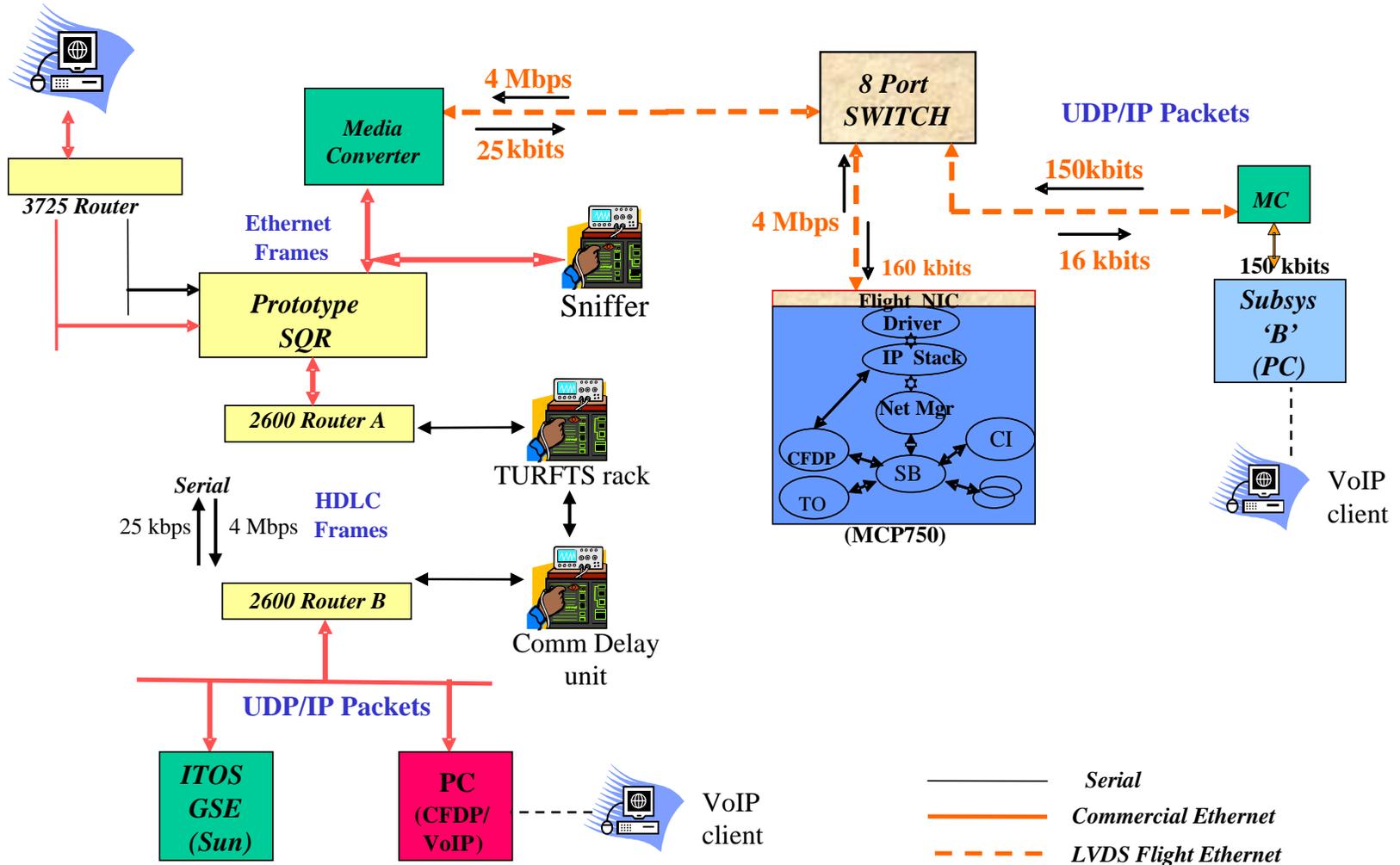
- **Plain-text Mode**
 - Simple packet forwarding via static routes.
 - No QoS rules.
 - Rate-limit rules apply to aggregate traffic flowing through the interface to simulate throughput over a space link.
- **IP-Sec Mode**
 - Forward Link traffic is authenticated & encrypted using IKE & IP-Sec via an IP-Sec tunnel between the Ground Station router and SQR.
 - Return Link traffic is plain-text, traversing an IP tunnel between the SQR and Ground Station router. (Note: encrypting high rate Return Link traffic yields a prohibitive CPU footprint on the spacecraft).
 - The space link is composed of two IP tunnels, each used unidirectionally; the Cisco smartguys figured that one out.
 - No crypto session timeouts so session keys are retained while out of contact, and when uni-directional links are operating.
 - 3DES & AES algorithms were used in conformance to GSFC mandates.



Operational Scenario Info

- **IPSec used for all IP operations**
 - All forward link traffic is encrypted.
 - All traffic not decrypted by IPSec is dropped at the SQR via access list rules; only packets delivered by the IPSec tunnel are forwarded.
 - Encrypted blind commanding supported.
 - Initial IKE handshake requires bidirectional comms. Once the sessions are established, the traffic composition is unchanged but contents are encrypted.
- **Due to lack of an IOS driver for a CPCI HDLC card in the SQR cage, an additional Cisco 2600 was used to convert the serial HDLC space-link to Ethernet**
 - no rate limiting, tunneling or QoS semantics were configured- only packet forwarding.
 - CPU impact on router from HDLC framing/deframing and link management was not testable.
 - IPSec is passed through; All tunnels extend from the SQR to the ground router.

Embedded Space Router Testbed





Questions to be answered:

- *CPU & memory utilization figures*
 - How busy is the cpu?
 - How much buffer memory is required to handle the traffic load?
- *Consequences of protocols such as IP-Sec, CFDP & TCP*
 - How do failures manifest?
 - How do protocols recover from partial or complete link failure?
 - How do protocols handle uni-directional link states & changes from uni-directional to bi-directional?
- *Management of the space-link.*
 - How does packet loss affect operations?



Results

- Had sufficient memory: no packet drops were recorded in the SQR.
- Bulk data transmission from Spacecraft to Mission Control w/ file transfer protocol responses traversing the IPSec tunnel: CPU footprint ~10%
- Same bulk transmission with bi-directional VoIP between Spacecraft and Mission Control: CPU footprint ~15% - VoIP forward traffic over the IPSec link was somewhat expensive.
- CFDP worked well
 - Unidirectional links work as expected; protocol handled the link transitions gracefully using the configured timeouts; no link state knowledge was incorporated in the protocol operation.
 - Protocol resumes and finishes without error as bidirectional comms are restored.
- IPSec is more complex
 - Once IKE sessions are established, crypto worked well unidirectionally & was a nice way to secure the command link without adding flight software.
 - Crypto time-out requires bi-directional link to initialize.
 - Cross-vendor compatibility issues; Sun vs Cisco.
 - Flight deployment will require significant management by Flight Software.
 - Need for pre-shared keys for contingency commanding.
- For general support of onboard TCP applications, a SCPS gateway is likely required for lunar scale delay products.
- Skype was annoying; configurable data-rates, uni-directional and “connection-less” calls are important features.
- IP as end-to-end transport was a big win; when the testbed was disassembled, moved, reassembled and integrated with the TURFTS system, only a couple hours of negotiating IP addresses and router config was necessary to bring up the entire system.



Conclusions (1 of 2)

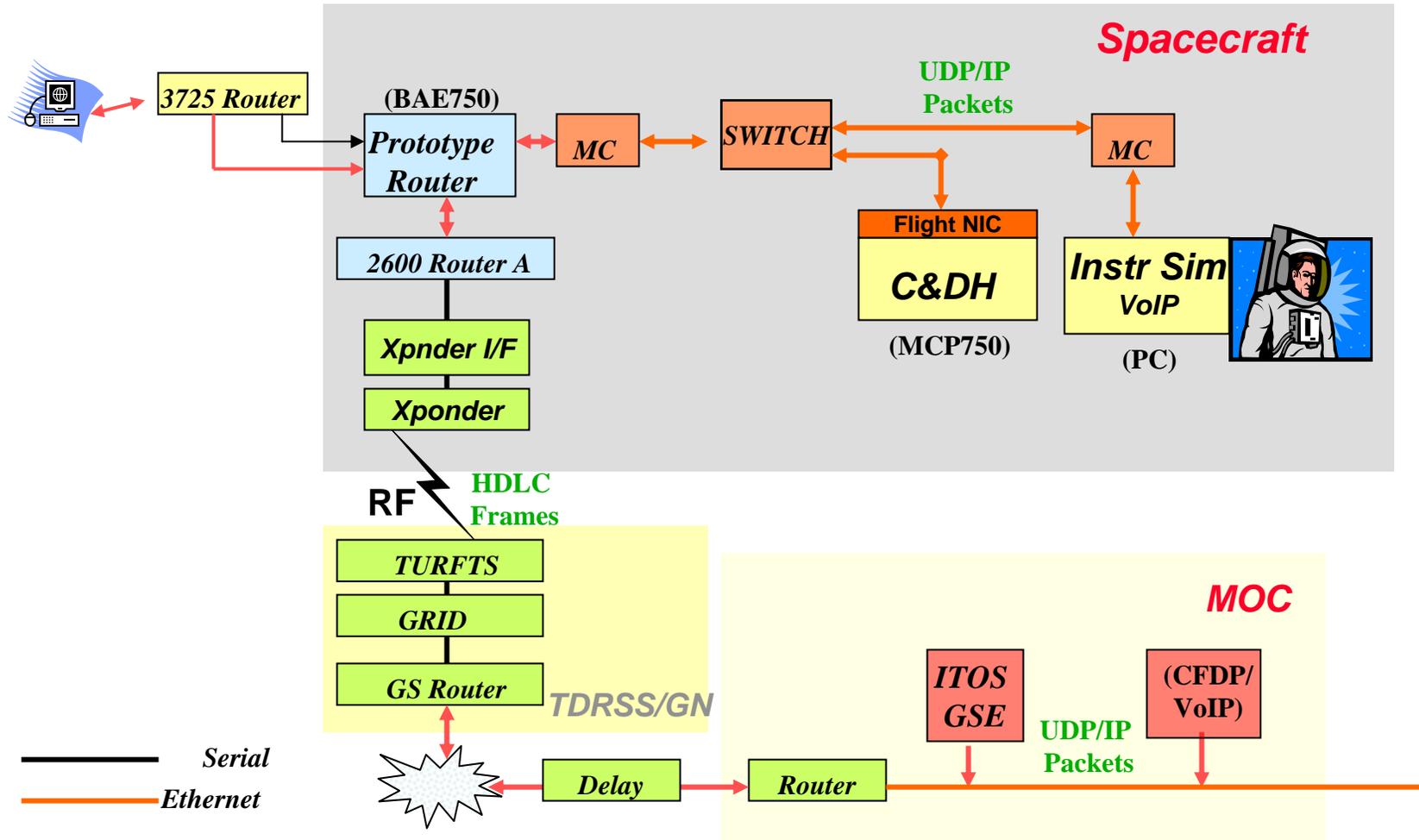
- ***Hardware footprint & performance***
 - Pro: Cisco SQR worked nicely as a standard IOS; entirely conformant with expected characteristics.
 - Con: Requires hardware resources roughly equivalent to a C&DH computer- but could scale down a bit before performance is considerably affected.
- ***Router command and control***
 - Command, control and monitoring features of IOS are sufficient for normal operation; SNMP and other tools are likely more than adequate.
 - Contingency modes will likely require a console serial port always operable & connected to the C&DH systems for command/control.
 - Deployed system would probably include digital I/O to reset, power on/off and configure SQR, and possibly firmware augmentation to parse magic command packets.
 - Onboard flash boot & maintenance features will be necessary.



Conclusions (2 of 2)

- *IP Quality of Service policies*
 - Forwarding policies at router must be carefully designed. Typical packet forwarding rules will not adequately control queuing & utilization over the space link.
 - Access-lists are needed at each end of the space link to ensure only valid space-link traffic traverses either link.
 - Application layer software must throttle itself to keep from overusing bandwidth and to conform to system engineering policy, particularly when transmitting over the space-link.
 - This is problematic when using IP because such measures are not usually implemented in the app layer at all. A feedback mechanism from the router feeding the space link is clearly required if the space link is to meet utilization requirements.
 - DiffServe/IntServe are helpful but insufficient as the sole means of implementing system engineering.
- *IP-Sec*
 - Disabling session timeouts is a cheap & easy way to obtain persistent state for long-term crypto.
 - Pre-shared, static keys for contingency operation should be considered.
 - IP-Sec CPU overhead makes encryption expensive for high rate Return Links. For the typical low bandwidth Forward Link, IP-Sec adds little cost and considerable security.

Integrated Ethernet/IP Testbed





Path to Flight

- **Embedded Space Router (ESR)**
 - BAE750 has compatible flight board
 - Add IOS driver for HDLC card.
 - Flash/EEPROM for IOS image (currently booting from outside source)
 - Implementation of command/control features for contingency ops.
- **NIC and Switch**
 - Flyable with rad-hard part upgrades & ETU board development.
 - Qualify/test to 100Mb- not precluded by existing design.
 - 100 megabit Switch may have reduced or mixed rate ports because of bandwidth limitations in the FPGA & memory.
 - Upgrade to 1 Gig would require a major revisit to the design.



For more information:

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